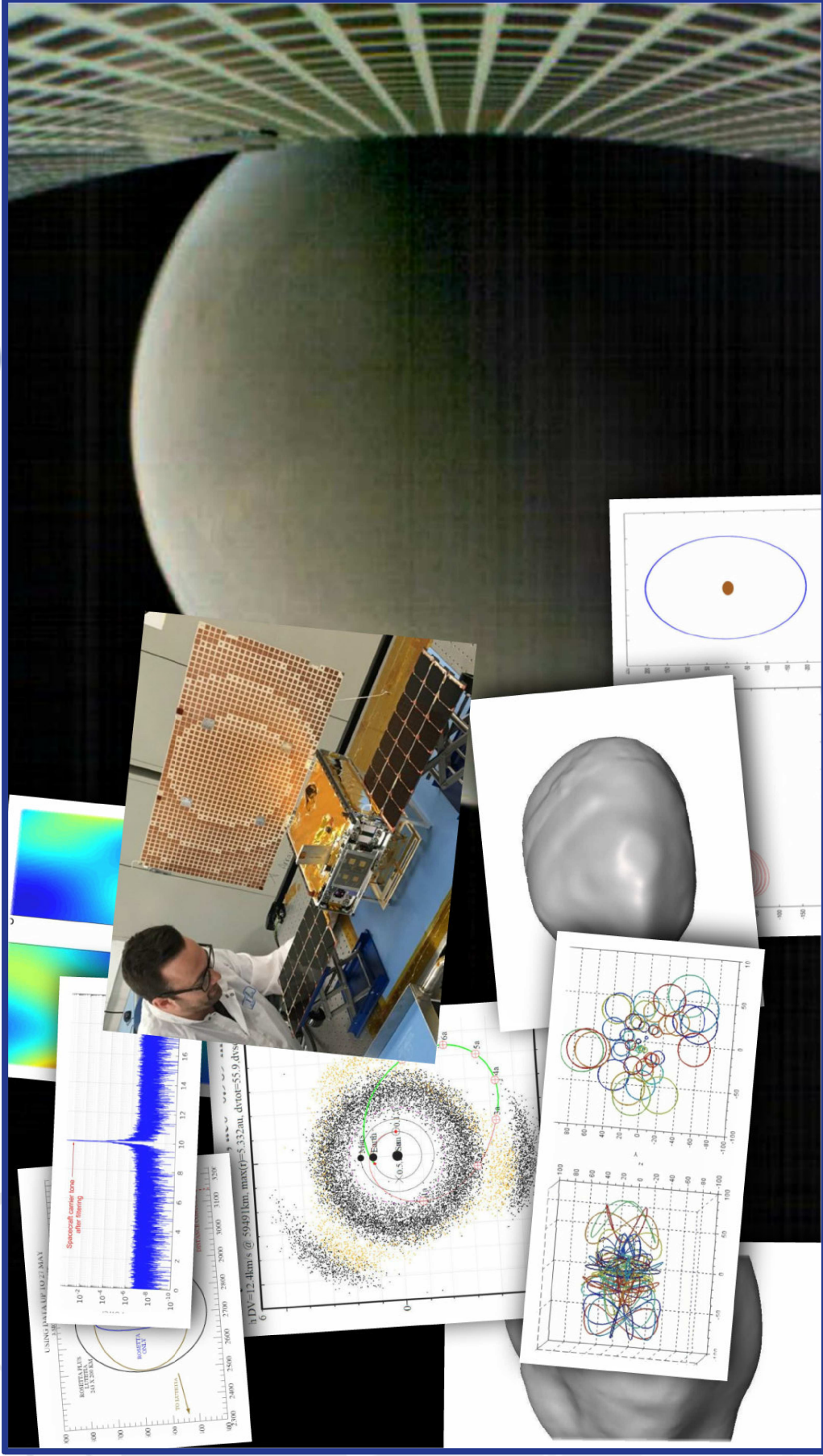


# DOCKS, a growing software suite for space mission prototyping

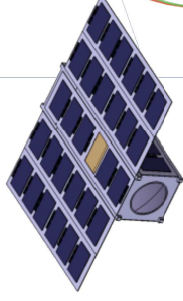
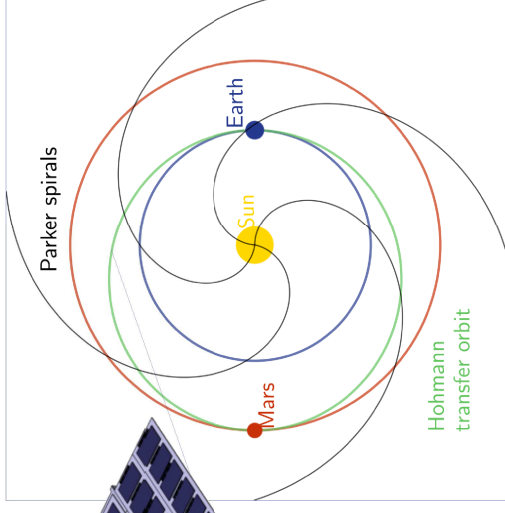
Boris Segret, Sebastien Durand  
*C<sup>2</sup>ERES, the space pole of PSL University Paris*

3<sup>rd</sup> OSCW, Athens, 14-16 oct. 2019

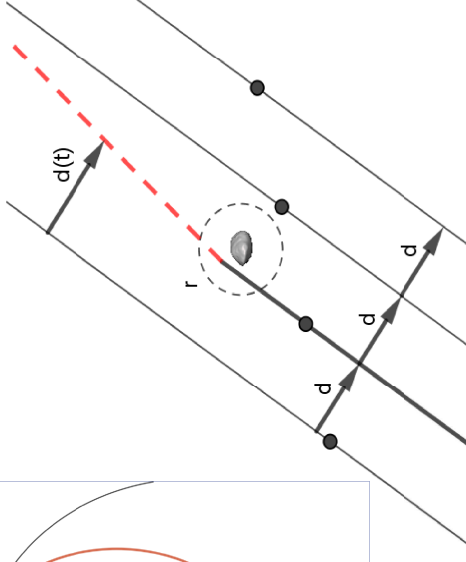
---



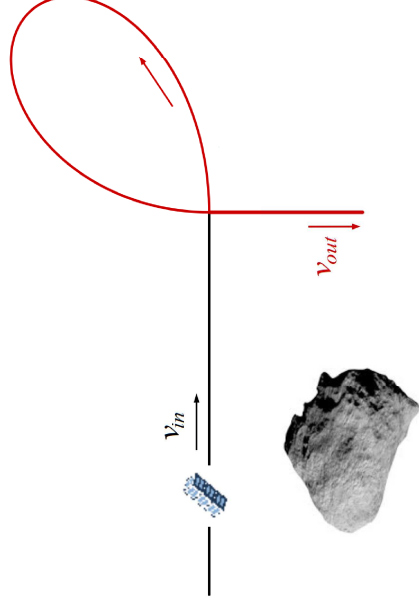
sources :  
 NASA MarCO  
 JIVE VLBI  
 ESA ROSETTA  
 CNES MMX  
 Etude NOIRE  
 MAT  
 Celestia/Motherlode

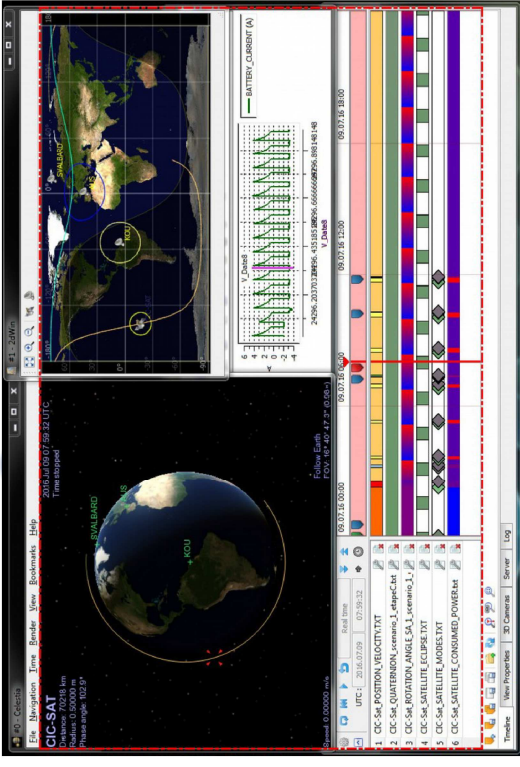


(Guinac, Segret)



| Earth   | Guiana |
|---|--------|
| 2023 Jun 13 19:54:09 UTC                              |        |
| Time stopped  |        |
| Altitude: 657960 km                                   |        |
| Radius: 6378.1 km                                     |        |
| Apogee distance: 4° 29' 42.3"                         |        |
| Perigee distance: 2.9°                                |        |
| Speed: 0.00000 m/s                                    |        |
| Earth: Earth  |        |
| POW: 5° 50' 31.8" (2.884)                             |        |
| Latitude = 59.9566848324 ; Longitude = -70.1882723813 |        |





CNES

ascii formats to VTS © CNES

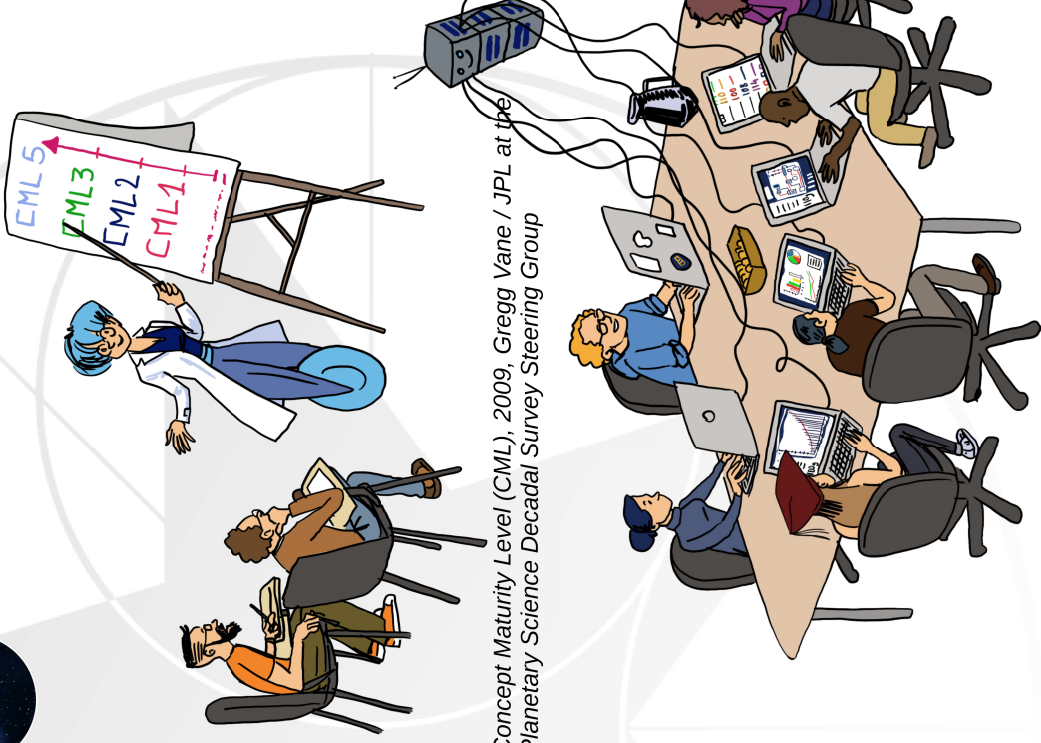


DOCKS helps and structures your CubeSat project, for you to focus:

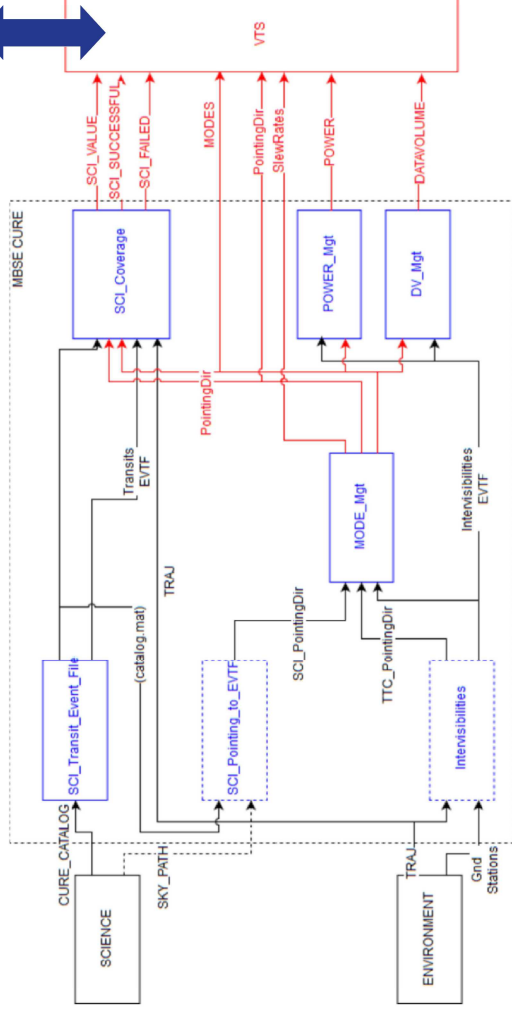
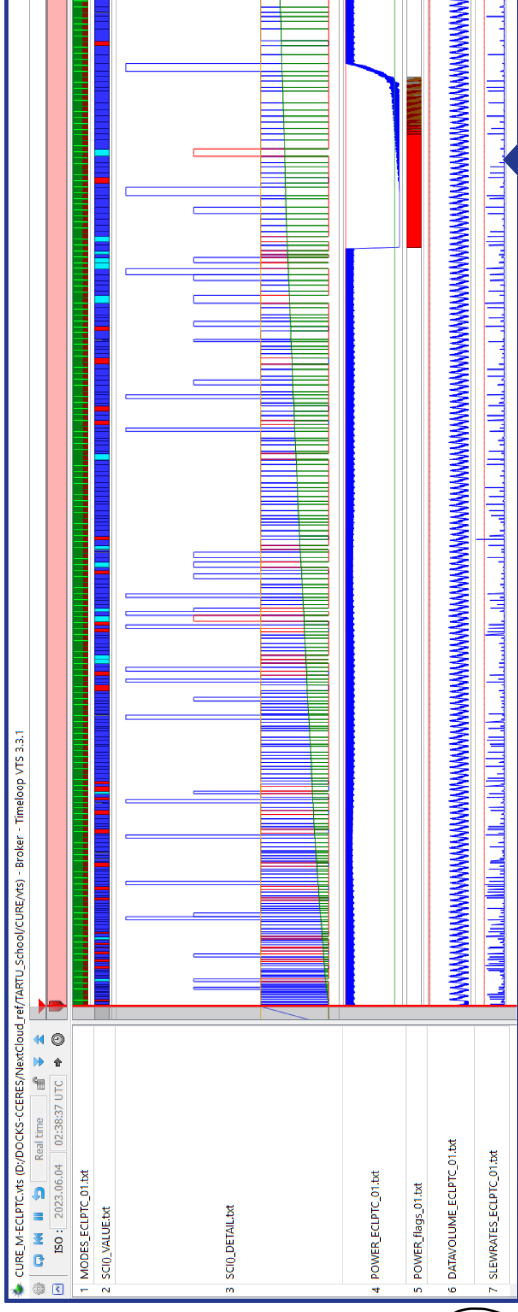
- ★ on the scientific coverage
- ★ on the engineering sizing

... from the early design up to AIT/AIV



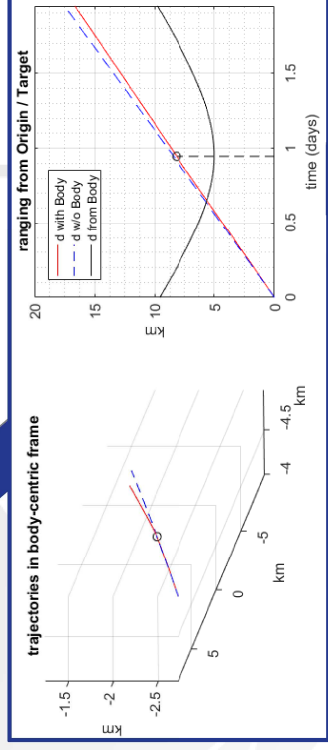


Concept Maturity Level (CML), 2009, Gregg Vane / JPL at the Planetary Science Decadal Survey Steering Group

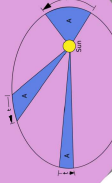




Propagator



CONIC



## Deep space trajectories

- ★ Cruise / RdV / ProxOps
  - ▶ Forward / Backward propagation
  - ▶ Adaptive time-step
    - Runge-Kutta or Runge-Kutta-Fehlberg
    - Accuracy-driven adaptive step
  - ▶ **Continuous propulsion**
- ★ Proximity Operations
  - ▶ Spherical Harmonics or other
  - ▶ Cross-check scientific models
  - ▶ Drag / comets...

## ★ Validations

- ▶ Based on planets and asteroids
- ▶ **Still some issues** in special contexts
- ▶ **Cross-checking welcome!**

Earth-vicinity: link to CNES' STELA

Keplerian engine: "CONIC"



Intervisibilities with....

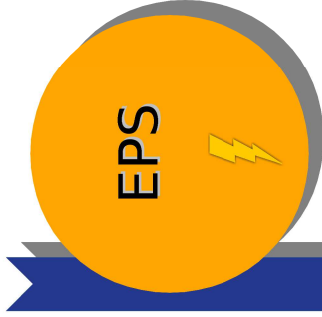
- ★ Sun, Ground station(s)
- ★ Output = “Event File” (EVTF)
- ★ Adaptive time accuracy



```

EVTF_INTERVISIBILITY_ECLPTC_01.txt
1  CIC MEM VERS = 1.0
2  CREATION_DATE = 2019-07-31T22:06:14.953
3  ORIGINATOR = DOCKS / CCERES / LabEx ESEP
   - Paris Observatory - PSL University Paris
4
5  META_START
6
7  COMMENT = Intervisibility for orbit :
   TRAJ_Ecliptic_01.txt, resampled at
   10-second intervals
8
9  USER_DEFINED_PROTOCOL = NONE
10 USER_DEFINED_CONTENT = CHRONOGRAMS
11 USER_DEFINED_SIZE = 1
12 USER_DEFINED_TYPE = STRING
13 USER_DEFINED_UNIT = [n/a]
14 TIME_SYSTEM = UTC
15
16 META_STOP
17
18 60096 2950.0 ECLIN00001
19 60096 3010.0 COMIN00011
20 60096 3600.0 COMEG00011
21 60096 5090.0 ECLEGG00001
22 60096 8740.0 ECLIN00001
23 60096 10870.0 ECLEGG00001
  
```

EVTF



EPS module

- ★ Solar arrays mounting, cells & battery techno
- ★ Mode strategy
- ★ Quaternions & Intervisibilities
- ★ GUI

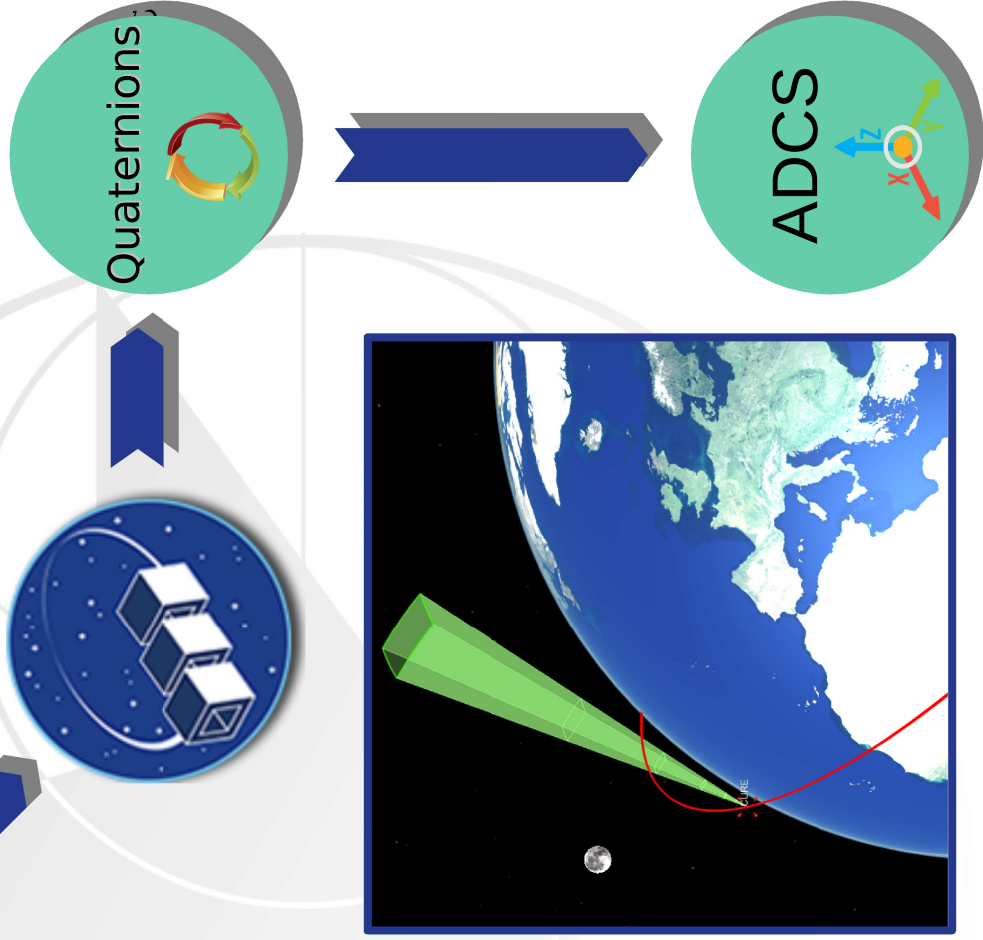
Datalink (to come)

- ★ Primitive modeling with intervisibilities
- ★ Mode strategy
- ★ Output = “Datavolume on board”
- ★ Quaternions (if relevant), Atmospheric loss



Datalink

Done  
To be done



## “Easy-Quaternions” module

- ★ (to be released)
- ★ 3-degree simple pointing strategies
- ★ Output
  - ▶ “directions” or “quaternions” in VTS formats
  - ▶ slew-rates

## “Non-easy quaternions” module

- ★ 4<sup>th</sup> degree of freedom
- ★ Coupling with Mode strategy

## Generic ADCS simulator

- ★ Not “Earth-limited”
- ★ Standard control laws and filtering to be tuned
- ★ Library of known actuators and sensors
- ★ Coupling with continuous propulsion



## What is done yet

- ★ Free licenses
- ★ Modular architecture
  - ▶ Python 3.x / Anaconda 3
    - astropy / polyastro / calceph
  - ▶ “docks-tools” module
- ★ Multi-OS
  - ▶ Ubuntu 18.04
  - ▶ MS-Windows 10
- ★ Distribution
  - ▶ public Gitlab
  - ▶ releases with install routines
  - ▶ support by mail

## What we will do next

- ★ Multi-OS:
    - ▶ Mac OS & Debian
    - ▶ Intensive testing in all OS
  - ★ Trajectories
    - ▶ Sensitivity analyses => Accuracy-driven
    - ▶ Propulsion arcs
  - ★ Energy: Initial GUI
  - ★ MBSE management
    - ▶ Timeline, Links and updates in the models
    - ▶ User’s configuration management
  - ★ “drop & run” service for heavy computation
    - ▶ Propagations (~50 it./s, per basic perturbation)
    - ▶ ADCS
- (.../....)

| Skills                 | A - Top Priority                            | B - High Priority                    | C - Interesting!            |
|------------------------|---|--------------------------------------|-----------------------------|
| USER                   | Report bugs                                 | Report boring use                    | Request features            |
| Scientist / Engineer   | Challenge Propagator's or EPS' performances | Add a model (Propagator or Datalink) | Module for Thermal analysis |
| IT enthusiast          | User-friendly GUI                           | Reading from GMAT/STK                | Wizards                     |
| Open Source enthusiast | Tutorials & Citations                       | "Modules' datasheet"                 | Templates                   |

★ Contact / Support (Gitlab): [docks.contact@obspm.fr](mailto:docks.contact@obspm.fr)

★ Partnering: [cceres.psl@obspm.fr](mailto:cceres.psl@obspm.fr)

★ Keep in touch: subscribe to “[cceres.new@obspm.fr](mailto:cceres.new@obspm.fr)”





# Thank you!

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★ Partnering: [cceres.psl@obspm.fr](mailto:cceres.psl@obspm.fr)

★ Keep in touch: subscribe to "[cceres.new@obspm.fr](mailto:cceres.new@obspm.fr)"





Propagator

Propagator: [About...]

Propagation Settings Numerical Methods Time Settings Output VTS Project

**Initial Conditions:**

Initial State Vector : [red bar]

Reference Frame Center : earth

Reference Frame : ICRF

Format

Time format JD

Position format KM

Velocity format KM/S

**Perturbations:**

Gravitational Ephemeris type imcce

Sun

Mercury

Venus

Earth Moon Barycenter

Mars System Barycenter

Jupiter System Barycenter

Saturn System Barycenter

Uranus System Barycenter

Neptune System Barycenter

Other [Manage]

**Complex gravitational model**

Adding complex gravitational model [Manage]

**Non-Gravitational**

Solar Radiation Pressure

OK Cancel Save Config Load Config Generate Clear all

Easy Trajectory

**CONIC**

Time Settings Orbital Elements Output

Orbital Center Body <custom>  Earth\_J2 0

Custom Body Name : Didymos mu (km3/s2) : 1,000000000 m3/s2

Trajectory File (ICRF-SSB) : [...]

Semi-Major Axis 150 km

Eccentricity 0,0000

Inclination (ICRF) 0,0000 deg

Longitude of the ascending node 0,0000 deg

(ICRF)

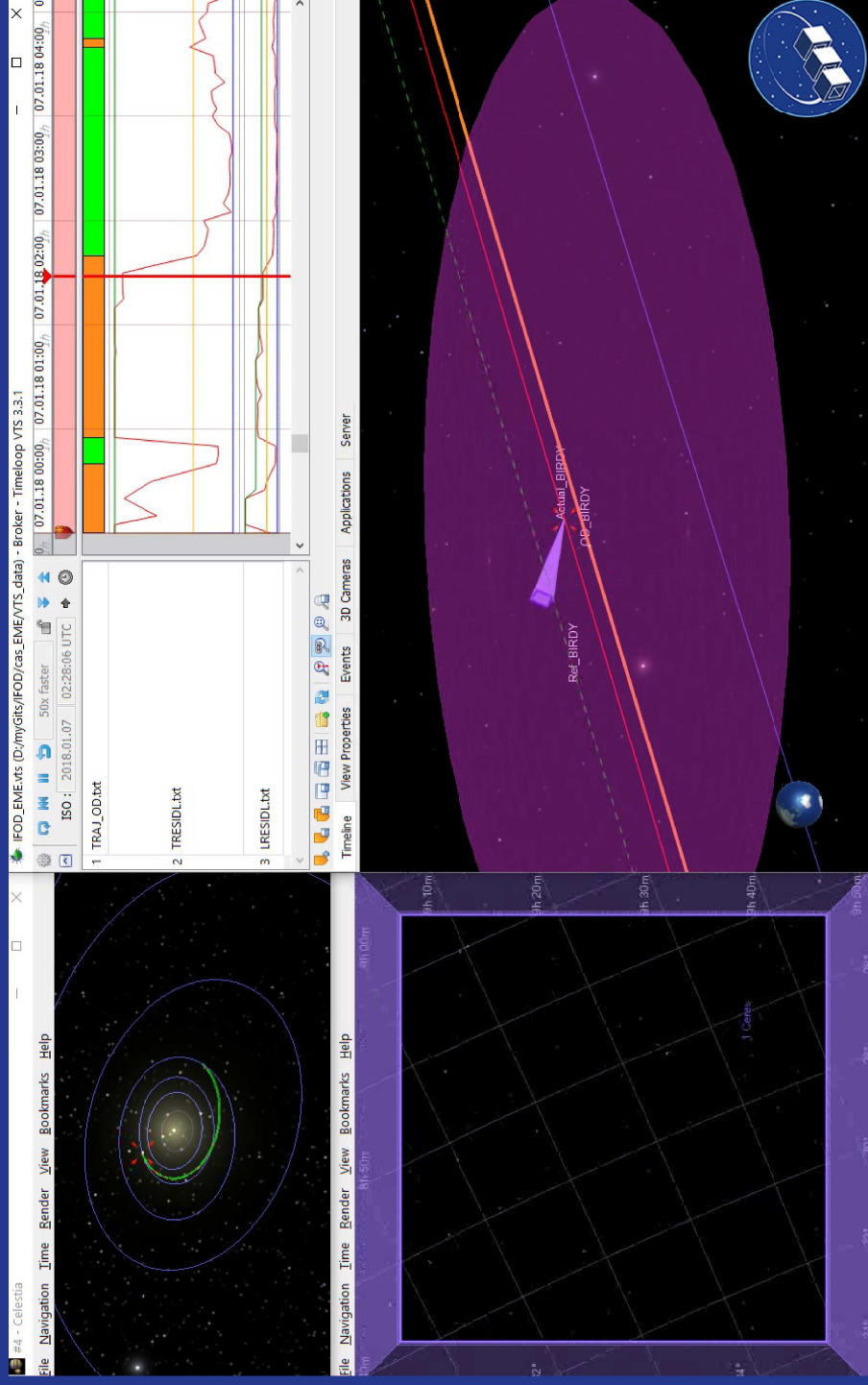
Argument of periapsis (ICRF) 0,0000 deg

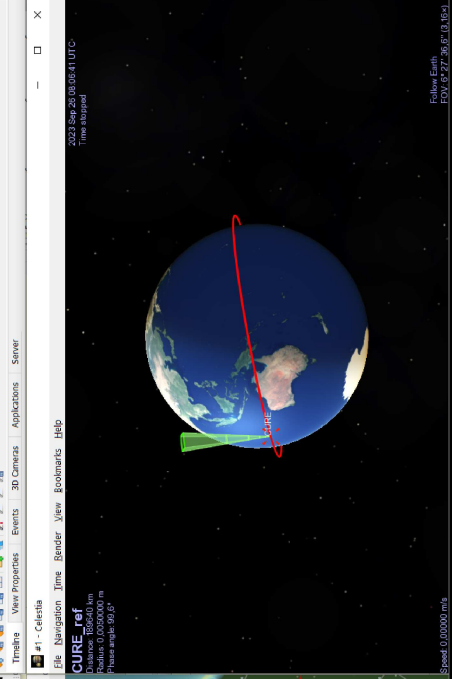
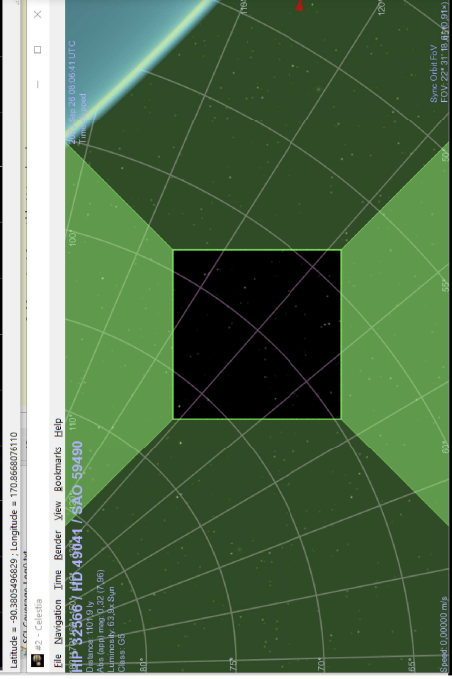
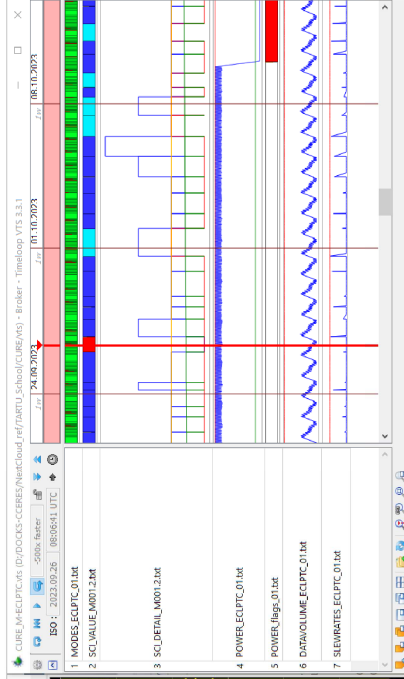
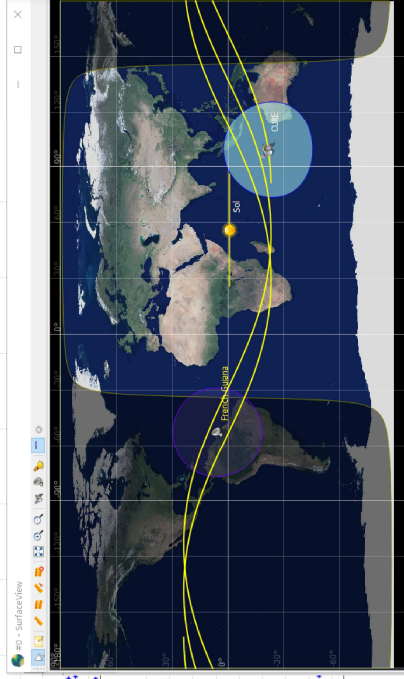
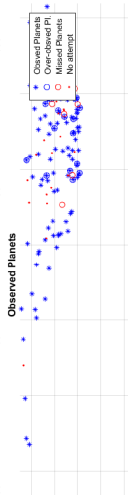
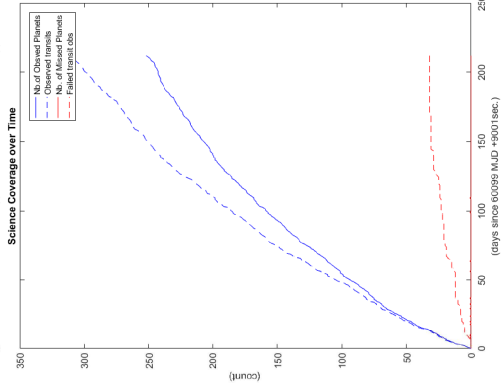
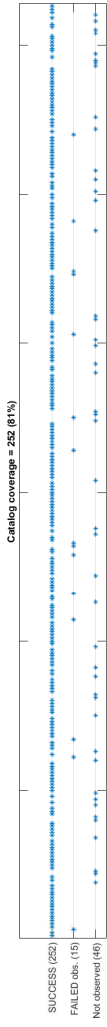
True anomaly at Start 0,0000 deg

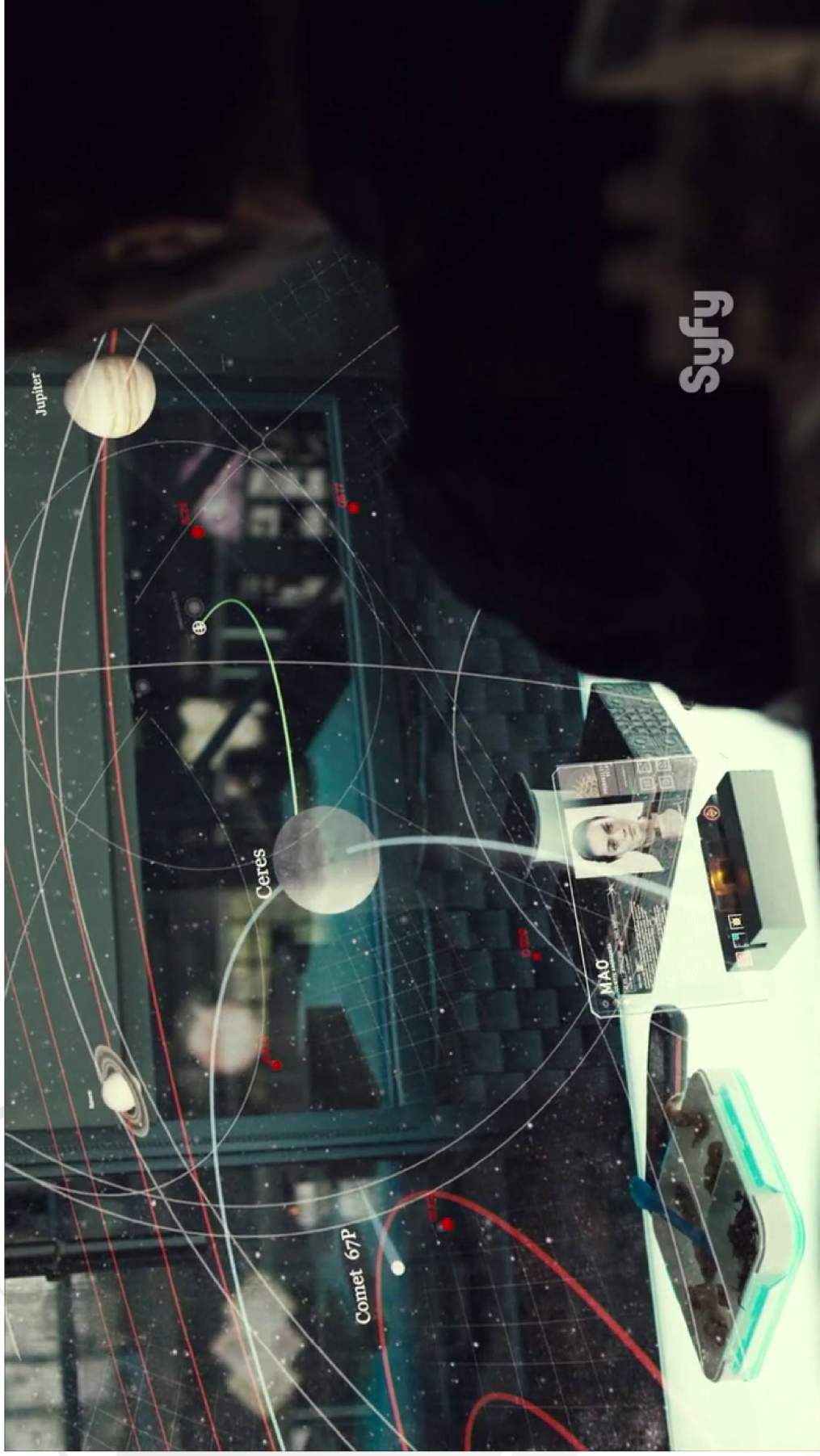
Clear All Generate Save Config Load Config OK Cancel

# Récapitulatif : UKF, 3 objets d'avant-plan

- ✧ Départ de la Terre
- ✧ Milieu de croisière
- ✧ Arrivée à Mars







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